ANALOG DEVICES

Four Degrees of Freedom Inertial Sensor

ADIS16300

FEATURES

14-bit digital gyroscope with digital range scaling ±75°/sec, ±150°/sec, ±300°/sec settings Tri-axis, 14-bit digital accelerometer ±3 g measurement range 13-bit pitch and roll incline calculations 330 Hz bandwidth 150 ms start-up time Factory-calibrated sensitivity, bias, and axial alignment **Digitally controlled bias calibration** Digitally controlled sample rate, up to 819.2 SPS External clock input enables sample rates up to 1200 SPS **Digitally controlled filtering** Programmable condition monitoring Auxiliary digital input/output **Digitally activated self-test** Programmable power management **Embedded temperature sensor** SPI-compatible serial interface Auxiliary, 12-bit ADC input and DAC output Single-supply operation: 4.75 V to 5.25 V 2000 g shock survivability Operating temperature range: -40°C to +85°C

FUNCTIONAL BLOCK DIAGRAM AUX AUX ADC DAC TEMPERATURE SENSOR cs MEMS SCLK ANGULAR RATE SENSOR CALIBRATION OUTPUT DIN AND DIGITAL REGISTERS AND SPI SIGNAL CONDITIONING DOUT PROCESSING AND CONVERSION TRI-AXIS MEMS CELERATION SENSOR ALARMS POWER vcc MANAGEMENT DIGITAL CONTROL SELF-TEST GND Ą ADIS16300 7842-001 RST DIO1 DIO2 DIO3 DIO4 Fiaure 1.

APPLICATIONS

Medical instrumentation Robotics Platform control Navigation

GENERAL DESCRIPTION

The ADIS16300 *i*Sensor[®] is a complete inertial system that includes a yaw rate gyroscope and tri-axis accelerometer. Each sensor in the ADIS16300 combines industry-leading *i*MEMS[®] technology with signal conditioning that optimizes dynamic performance. The factory calibration characterizes each sensor for sensitivity, bias, alignment, and linear acceleration (gyro bias). As a result, each sensor has its own dynamic compensation for correction formulas that provide accurate sensor measurements over the specified power supply range of +4.75 V to +5.25 V. The ADIS16300 provides a simple, cost-effective method for integrating accurate, multi-axis, inertial sensing into industrial systems, especially when compared with the complexity and investment associated with discrete designs. All necessary motion testing and calibration are part of the production process at the factory, greatly reducing system integration time. Tight orthogonal alignment simplifies inertial frame alignment in navigation systems. An improved SPI interface and register structure provide faster data collection and configuration control. The ADIS16300, along with a flex interface, drops into current systems that use the ADIS1635x family, providing the opportunity to scale cost for systems that only require four degrees of freedom inertial sensing. This compact module is approximately 23 mm × 31 mm × 7.5 mm and provides a standard connector interface, which enables horizontal or vertical mounting.

Rev. A

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REVISION HISTORY

4/09—Rev. 0 to Rev. A

Changes to Figure 5 and Figure 6	7
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10/08—Revision 0: Initial Version	

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SPECIFICATIONS

 $T_A = -40^{\circ}$ C to +85°C, VCC = 5.0 V, angular rate = 0°/sec, dynamic range = ±300°/sec, ±1 g, unless otherwise noted.

Table 1.

Parameter	Test Conditions	Min	Тур	Мах	Unit
GYROSCOPE					
Dynamic Range		±300	±375		°/sec
Initial Sensitivity	$T_A = 25^{\circ}C$, dynamic range = $\pm 300^{\circ}$ /sec	0.0495	0.05	0.0505	°/sec/LSB
	$T_A = 25^{\circ}C$, dynamic range = $\pm 150^{\circ}$ /sec		0.025		°/sec/LSB
	$T_A = 25^{\circ}C$, dynamic range = $\pm 75^{\circ}$ /sec		0.0125		°/sec/LSB
Sensitivity Temperature Coefficient			400		ppm/°C
Misalignment	Reference to z-axis accelerometer, $T_A = 25^{\circ}C$		0.1		Degrees
	Axis-to-frame (package), $T_A = 25^{\circ}C$		±0.5		Degrees
Nonlinearity	Best fit straight line		0.1		% of FS
Initial Bias Error	$T_A = 25^{\circ}C, \pm 1 \sigma$		±3		°/sec
In-Run Bias Stability	T _A = 25°C, 1 σ, SMPL_PRD = 0x01		0.007		°/sec
Angular Random Walk	$T_A = 25^{\circ}C$, 1 σ , SMPL_PRD = 0x01		1.9		°/√hr
Bias Temperature Coefficient			0.1		°/sec/°C
Linear Acceleration Effect on Bias	Any axis, 1 σ (MSC_CTRL Bit [7] = 1)		0.05		°/sec/g
Voltage Sensitivity	VCC = 4.75 V to 5.25 V		0.25		°/sec/V
Output Noise	$T_A = 25^{\circ}C, \pm 300^{\circ}/sec$ range, no filtering		1.1		°/sec rms
Rate Noise Density	$T_A = 25^{\circ}C$, $f = 25$ Hz, $\pm 300^{\circ}$ /sec, no filtering		0.038		°/sec/√Hz rm
3 dB Bandwidth	· · · · · · · · · · · · · · · · · · ·		330		Hz
Sensor Resonant Frequency			14.5		kHz
Self-Test Change in Output Response	±300°/sec range setting	±696	±1400	±2449	LSB
ACCELEROMETERS	Each axis	1010	1100	±2117	250
Dynamic Range		±3	±3.6		0
Initial Sensitivity	25℃	0.594	<u>-</u> 3.0 0.6	0.606	<i>g</i> mg/LSB
Sensitivity Temperature Coefficient	X axis and Y axis	0.394	250	0.000	ppm/°C
sensitivity remperature coefficient	Z axis		250 300		
Micelianana					ppm/°C
Misalignment	Axis-to-axis, $T_A = 25^{\circ}$ C, $\Delta = 90^{\circ}$ ideal		±0.25		Degrees
NI 11 14	Axis-to-frame (package), $T_A = 25^{\circ}C$		±0.5		Degrees
Nonlinearity	Best fit straight line		±0.3		% of FS
Initial Bias Error	$T_A = 25^{\circ}C, \pm 1 \sigma, X \text{ axis and } Y \text{ axis}$		±60		mg
	$T_A = 25^{\circ}C, \pm 1 \sigma, Z axis$		±110		mg
In-Run Bias Stability	$T_A = 25^{\circ}C$, 1 σ , X axis and Y axis		0.048		mg
	$T_A = 25^{\circ}C$, 1 σ , Z axis		0.054		mg
Velocity Random Walk	$T_A = 25^{\circ}$ C, 1 σ , X axis and Y axis		0.118		m/sec/√hr
	$T_A = 25^{\circ}C$, 1 σ , Z axis		0.164		m/sec/√hr
Bias Temperature Coefficient	X axis and Y axis		2.5		m <i>g/</i> °C
	Z axis		4.5		mg/°C
Output Noise	$T_A = 25^{\circ}$ C, no filtering, X axis and Y axis		5		mg rms
	$T_A = 25^{\circ}$ C, no filtering, Z axis		7.5		mg rms
Noise Density	$T_A = 25^{\circ}$ C, no filtering, X axis and Y axis		0.2		mg/√Hz rms
	$T_A = 25^{\circ}C$, no filtering, Z axis		0.3		mg/√Hz rms
3 dB Bandwidth			330		Hz
Sensor Resonant Frequency			5.5		kHz
Self-Test Change in Output Response	X axis and Y axis	500	1100	1700	LSB
	Z axis	90	450	860	LSB
INCLINOMETER					
Sensitivity			0.044		°/LSB
					1
TEMPERATURE SENSOR					

Parameter	Conditions	Min	Тур	Max	Unit
ADC INPUT					
Resolution			12		Bits
Integral Nonlinearity			±2		LSB
Differential Nonlinearity			±1		LSB
Offset Error			±4		LSB
Gain Error			±2		LSB
Input Range		0		+3.3	V
Input Capacitance	During acquisition		20		pF
DAC OUTPUT	5 kΩ/100 pF to GND				
Resolution			12		Bits
Relative Accuracy	For Code 101 to Code 4095		±4		LSB
Differential Nonlinearity			±1		LSB
Offset Error			±5		mV
Gain Error			±0.5		%
Output Range		0		+3.3	V
Output Impedance			2		Ω
Output Settling Time			10		μs
LOGIC INPUTS ¹					
Input High Voltage, V _{INH}		2.0			V
Input Low Voltage, V _{INL}				0.8	V
	\overline{CS} signal to wake up from sleep mode			0.55	V
CS Wake-Up Pulse Width		20			μs
Logic 1 Input Current, I _{INH}	$V_{IH} = 3.3 V$		±0.2	±10	μΑ
Logic 0 Input Current, I _{INL}	$V_{IL} = 0 V$				
All Pins Except RST			-40	-60	μA
RST Pin			-1		mA
Input Capacitance, C _{IN}			10		pF
DIGITAL OUTPUTS ¹					
Output High Voltage, Vон	Isource = 1.6 mA	2.4			v
Output Low Voltage, Vo∟	$I_{SINK} = 1.6 \text{ mA}$			0.4	V
FLASH MEMORY	Endurance ²	10,000			Cycles
Data Retention ³	T_ = 85°C	20			Years
FUNCTIONAL TIMES ⁴	Time until data is available				
Power-On Start-up Time	Normal mode, SMPL_PRD \leq 0x09		180		ms
• -	Low power mode, SMPL_PRD \geq 0x0A		245		ms
Reset Recovery Time	Normal mode, SMPL_PRD \leq 0x09		55		ms
·	Low power mode, SMPL_PRD \geq 0x0A		120		ms
Sleep Mode Recovery Time			2.5		ms
Flash Memory Test Time	Normal mode, SMPL_PRD \leq 0x09		17		ms
-	Low power mode, SMPL_PRD \geq 0x0A		90		ms
Automatic Self-Test Time			12		ms
CONVERSION RATE	SMPL_PRD = 0x01 to 0xFF	0.413		819.2	SPS
Clock Accuracy		_		±3	%
Sync Input Clock				1.2	kHz
POWER SUPPLY	Operating voltage range, VCC	4.75	5.0	5.25	V
Power Supply Current	Low power mode at 25°C		18		mA
and the second second	Normal mode at 25°C		42		mA
	Sleep mode at 25°C		500		μA

¹ The digital I/O signals are driven by an internal 3.3 V supply, and the inputs are 5 V tolerant.
² Endurance is qualified as per JEDEC Standard 22, Method A117, and measured at -40°C, +25°C, and +125°C.
³ The retention lifetime equivalent is at a junction temperature (T₂) of 85°C as per JEDEC Standard 22, Method A117. Retention lifetime decreases with junction temperature.
⁴ These times do not include thermal settling and internal filter response times (330 Hz bandwidth), which may impact overall accuracy.

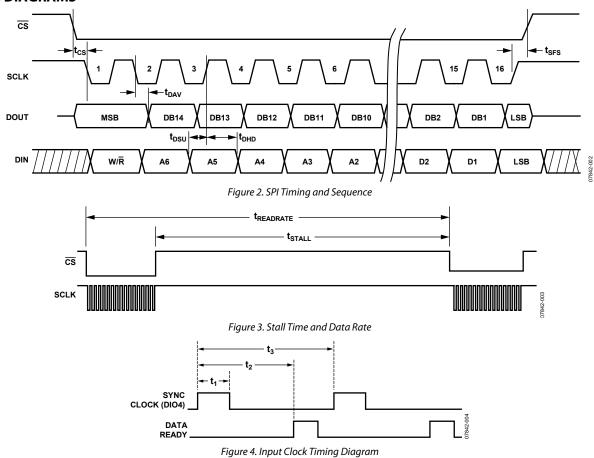
TIMING SPECIFICATIONS

 $T_A = 25^{\circ}$ C, VCC = 5 V, unless otherwise noted.

Table 2.

		-	ormal M L_PRD <u><</u>	ode <u><</u> 0x09)	_	Power N L_PRD <u>></u>		Bu	ırst Mo	de	
Parameter	Description	Min ¹	Тур	Max	Min ¹	Тур	Max	Min ¹	Тур	Max	Unit
f _{SCLK}		0.01		2.0	0.01		0.3	0.01		1.0	MHz
t stall	Stall period between data	9			75			1/f _{SCLK}			μs
t _{READRATE}	Read rate	40			100						us
tcs	Chip select to clock edge	48.8			48.8			48.8			ns
t _{DAV}	DOUT valid after SCLK edge			100			100			100	ns
t _{DSU}	DIN setup time before SCLK rising edge	24.4			24.4			24.4			ns
t DHD	DIN hold time after SCLK rising edge	48.8			48.8			48.8			ns
tsclkr, tsclkf	SCLK rise/fall times		5	12.5		5	12.5		5	12.5	ns
t _{DF} , t _{DR}	DOUT rise/fall times		5	12.5		5	12.5		5	12.5	ns
tsfs	CS high after SCLK edge	5			5			5			ns
t1	Input sync pulse width		5								μs
t ₂	Input sync to data ready output		600								μs
t ₃	Input sync period	833									μs

¹Guaranteed by design and characterization, but not tested in production.



TIMING DIAGRAMS

ABSOLUTE MAXIMUM RATINGS

Table 3.

Tuble 5.	
Parameter	Rating
Acceleration	
Any Axis, Unpowered	2000 g
Any Axis, Powered	2000 g
VCC to GND	–0.3 V to +6.0 V
Digital Input Voltage to GND	–0.3 V to +5.3 V
Digital Output Voltage to GND	-0.3 V to VCC + 0.3 V
Analog Input to GND	–0.3 V to +3.6 V
Operating Temperature Range	-40°C to +85°C
Storage Temperature Range	-65°C to +125°C ^{1, 2}

¹Extended exposure to temperatures outside the specified temperature range of -40° C to $+85^{\circ}$ C can adversely affect the accuracy of the factory calibration. For best accuracy, store the parts within the specified operating range of -40° C to $+85^{\circ}$ C.

² Although the device is capable of withstanding short-term exposure to 150°C, long-term exposure threatens internal mechanical integrity.

Stresses above those listed under Absolute Maximum Ratings may cause permanent damage to the device. This is a stress rating only; functional operation of the device at these or any other conditions above those indicated in the operational section of this specification is not implied. Exposure to absolute maximum rating conditions for extended periods may affect device reliability.

Table 4. Package Characteristics

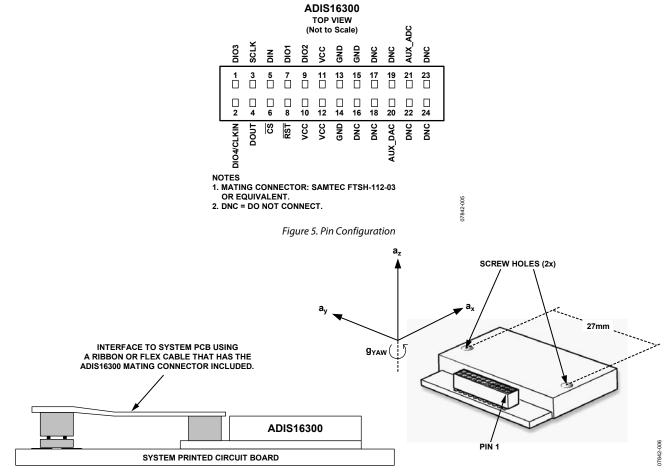
0			
Package Type	θ」Α	θις	Device Weight
24-Lead Module	39.8°C/W	14.2°C/W	16 grams

ESD CAUTION



ESD (electrostatic discharge) sensitive device. Charged devices and circuit boards can discharge without detection. Although this product features patented or proprietary protection circuitry, damage may occur on devices subjected to high energy ESD. Therefore, proper ESD precautions should be taken to avoid performance degradation or loss of functionality.

PIN CONFIGURATION AND FUNCTION DESCRIPTIONS



NOTES

1. ACCELERATION (a_X, a_Y, a_Z) AND ROTATIONAL (g_{YAW}) ARROWS INDICATE THE DIRECTION OF MOTION THAT PRODUCES A POSITIVE OUTPUT.

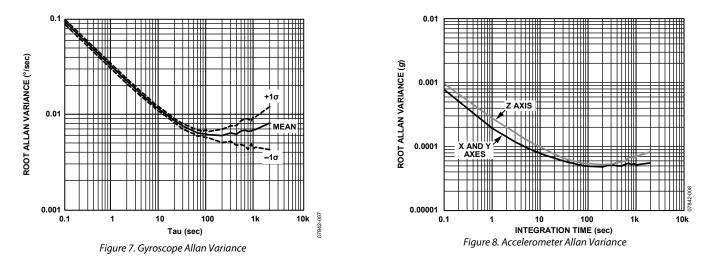
Figure 6. Device Orientation, Mounting, and Interface Diagrams

Table 5.	Pin Fu	inction	Descriptions
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Pin No.	Mnemonic	Type ¹	Description
1	DIO3	I/O	Configurable Digital Input/Output.
2	DIO4/CLKIN	I/O	Configurable Digital Input/Output or Sync Clock Input.
16, 17, 18, 19, 22, 23, 24	DNC	N/A	Do Not Connect.
3	SCLK	1	SPI Serial Clock.
4	DOUT	0	SPI Data Output. Clocks output on SCLK falling edge.
5	DIN	1	SPI Data Input. Clocks input on SCLK rising edge.
6	CS	I	SPI Chip Select.
7	DIO1	I/O	Configurable Digital Input/Output.
8	RST	1	Reset.
9	DIO2	I/O	Configurable Digital Input/Output.
10, 11, 12	VCC	S	Power Supply.
13, 14, 15	GND	S	Power Ground.
20	AUX_DAC	0	Auxiliary, 12-Bit DAC Output.
21	AUX_ADC	1	Auxiliary, 12-Bit ADC Input.

¹ S is supply, O is output, I is input, N/A is not applicable.

TYPICAL PERFORMANCE CHARACTERISTICS



BASIC OPERATION

The ADIS16300 is an autonomous sensor system that starts up after it has a valid power supply voltage and begins producing inertial measurement data at a sample rate of 819.2 SPS. After each sample cycle, the sensor data loads into the output registers and DIO1 pulses, providing a new-data-ready control signal for driving system-level interrupt service routines. In a typical system, a master processor accesses the output data registers through the SPI interface, using the hook-up shown in Figure 9. Table 6 provides a generic, functional description for each pin on the master processor. Table 7 describes the typical master processor settings normally found in a configuration register and used for communicating with the ADIS16300.

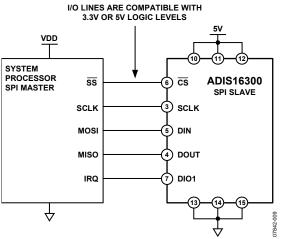


Figure 9. Electrical Hook-Up Diagram

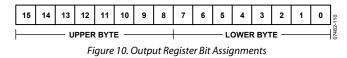
Table 6. Generic Master Processor Pin Names ar	d Functions
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Pin Name	Function
SS	Slave select
IRQ	Interrupt request
MOSI	Master output, slave input
MISO	Master input, slave output
SCLK	Serial clock

Table 7. Generic Master Processor SPI Settings

¹ For burst mode, SCLK rate \leq 1 MHz. For low power mode, SCLK rate \leq 300 kHz.

The user registers provide addressing for all input/output operations on the SPI interface. Each 16-bit register has two 7-bit addresses: one for its upper byte and one for its lower byte. Table 8 provides the lower-byte address for each register, and Figure 10 provides the generic bit assignments.



READING SENSOR DATA

Although the ADIS16300 produces data independently, it operates as an SPI slave device, which communicates with system (master) processors using the 16-bit segments displayed in Figure 11. Individual register reads require two 16-bit sequences. The first 16-bit sequence provides the read command bit (R/W = 0) and the target register address (A6...A0). The second sequence transmits the register contents (D15...D0) on the DOUT line. For example, if DIN= 0x0A00, then the content of XACCL_OUT shifts out on the DOUT line during the next 16-bit sequence.

The SPI operates in full duplex mode, which means that the master processor can read the output data from DOUT while using the same SCLK pulses to transmit the next target address on DIN.

DEVICE CONFIGURATION

The user register memory map (see Table 8) identifies configuration registers with either a W or R/W. Configuration commands also use the bit sequence displayed in Figure 12. If the MSB is equal to 1, the last eight bits (DC7...DC0) in the DIN sequence load into the memory address associated with the address bits (A5...A0). For example, if the DIN = 0xA11F, then 0x1F loads into Address Location 0x26 (ALM_MAG1, upper byte) at the conclusion of the data frame.

Most of the registers have a backup location in nonvolatile flash memory. The master processor must manage the backup function. Set GLOB_CMD[3] = 1 (DIN = 0xBE01) to execute a manual flash update (backup) operation, which copies the user registers into their respective flash memory locations. This operation takes 50 ms and requires the power supply voltage to be within the specified limit to complete properly. The FLASH_CNT register provides a running count of these events for managing the long-term reliability of the flash memory.

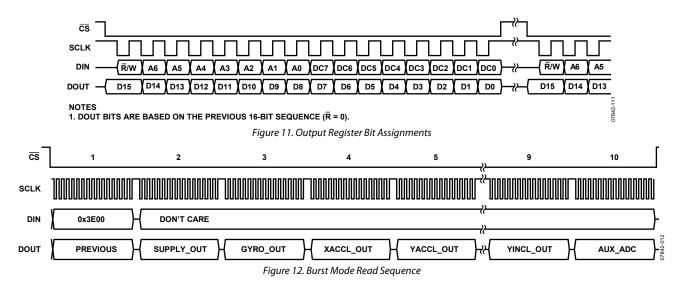
BURST MODE DATA COLLECTION

Burst mode data collection offers a more process-efficient method for collecting data from the ADIS16300. In 10 sequential data cycles (each separated by one SCLK period), all nine output registers clock out on DOUT. This sequence starts when the DIN sequence is 0011 1110 0000 0000 (0x3E00). Next, the contents of each output register are output from DOUT, starting with SUPPLY_OUT and ending with AUX_ADC (see Figure 12). The addressing sequence shown in Table 8 determines the order of the outputs in burst mode.

Table 8. User Register Memory Map						
Name	R/W	Flash Backup	Address ¹	Default	Function	Bit Assignments
FLASH_CNT	R	Yes	0x00	N/A	Flash memory write count	N/A
SUPPLY_OUT	R	No	0x02	N/A	Power supply measurement	Table 9
GYRO_OUT	R	No	0x04	N/A	X-axis gyroscope output	Table 9
N/A	N/A	N/A	0x06	N/A	Reserved	N/A
N/A	N/A	N/A	0x08	N/A	Reserved	N/A
XACCL_OUT	R	No	0x0A	N/A	X-axis accelerometer output	Table 9
YACCL_OUT	R	No	0x0C	N/A	Y-axis accelerometer output	Table 9
ZACCL_OUT	R	No	0x0E	N/A	Z-axis accelerometer output	Table 9
TEMP_OUT	R	No	0x10	N/A	X-axis gyroscope temperature measurement	Table 9
PITCH_OUT	R	No	0x12	N/A	X-axis inclinometer output measurement	Table 9
ROLL_OUT	R	No	0x14	N/A	Y-axis inclinometer output measurement	Table 9
AUX_ADC	R	No	0x16	N/A	Auxiliary ADC output	Table 9
N/A	N/A	N/A	0x18	N/A	Reserved	N/A
GYRO_OFF	R/W	Yes	0x1A	0x0000	X-axis gyroscope bias offset factor	Table 10
N/A	N/A	N/A	0x1C	N/A	Reserved	N/A
N/A	N/A	N/A	0x1E	N/A	Reserved	N/A
XACCL_OFF	R/W	Yes	0x20	0x0000	X-axis acceleration bias offset factor	Table 11
YACCL_OFF	R/W	Yes	0x22	0x0000	Y-axis acceleration bias offset factor	Table 11
ZACCL_OFF	R/W	Yes	0x24	0x0000	Z-axis acceleration bias offset factor	Table 11
ALM_MAG1	R/W	Yes	0x26	0x0000	Alarm 1 amplitude threshold	Table 22
ALM_MAG2	R/W	Yes	0x28	0x0000	Alarm 2 amplitude threshold	Table 22
ALM_SMPL1	R/W	Yes	0x2A	0x0000	Alarm 1 sample size	Table 23
ALM_SMPL2	R/W	Yes	0x2C	0x0000	Alarm 2 sample size	Table 23
ALM_CTRL	R/W	Yes	0x2E	0x0000	Alarm control	Table 24
AUX_DAC	R/W	No	0x30	0x0000	Auxiliary DAC data	Table 18
GPIO_CTRL	R/W	No	0x32	0x0000	Auxiliary digital input/output control	Table 16
MSC_CTRL	R/W	Yes	0x34	0x0006	Miscellaneous control	Table 17
SMPL_PRD	R/W	Yes	0x36	0x0001	Internal sample period (rate) control	Table 13
SENS_AVG	R/W	Yes	0x38	0x0402	Dynamic range/digital filter control	Table 15
SLP_CNT	W	No	0x3A	0x0000	Sleep mode control	Table 14
DIAG_STAT	R	No	0x3C	0x0000	System status	Table 21
GLOB_CMD	W	N/A	0x3E	0x0000	System command	Table 12

Table 8. User Register Memory Map

¹ Each register contains two bytes. The address of the lower byte is displayed. The address of the upper byte is equal to the address of the lower byte, plus 1.



OUTPUT DATA REGISTERS

Figure 6 provides the positive measurement direction for each inertial sensor (gyroscope and accelerometers). Table 9 provides the configuration and scale factor for each output data register in the ADIS16300. All inertial sensor outputs are 14-bits in length and are in twos complement format, which means that 0x0000 is equal to 0 LSB, 0x0001 is equal to +1 LSB, and 0x3FFF is equal to -1 LSB. The following is an example of how to calculate the sensor measurement from the GYRO_OUT:

 $GYRO_OUT = 0x3B4A$

 $0x000 - 0x33B4A = -0x04B6 = -(4 \times 256 + 11 \times 16 + 6)$

$$-0x04B6 = -1206$$
 LSB

Rate = $0.05^{\circ}/\sec \times (-1206) = -60.3^{\circ}/\sec$

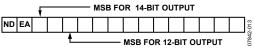
Therefore, a GYRO_OUT output of 0x3B4A corresponds to a clockwise rotation about the z-axis (see Figure 6) of 60.3°/sec when looking at the top of the package.

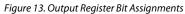
Table 9. Output Data Register Formats

Register	Bits	Format	Scale	
SUPPLY_OUT	12	Binary, 5 V = 0x0814	2.42 mV	
GYRO_OUT ¹	14	Twos complement	0.05°/sec	
XACCL_OUT	14	Twos complement	0.6 m <i>g</i>	
YACCL_OUT	14	Twos complement	0.6 m <i>g</i>	
ZACCL_OUT	14	Twos complement	0.6 m <i>g</i>	
TEMP_OUT	12	Twos complement 25°C = 0x0000	0.14°C	
ROLL_OUT	13	Twos complement	0.044°	
PITCH_OUT	13	Twos complement	0.044°	
AUX_ADC	12	Binary, 1 V = 0x04D9	0.81 mV	

 $^{\rm 1}$ Assumes that the scaling is set to \pm 300°/sec. This factor scales with the range.

Each output data register uses the bit assignments shown in Figure 13. The ND flag indicates that unread data resides in the output data registers. This flag clears and returns to 0 during an output register read sequence. It returns to 1 after the next internal sample updates the registers with new data. The EA flag indicates that one of the error flags in the DIAG_STAT register (see Table 21) is active (true). The remaining 14-bits are for data.





Inclinometers

The ROLL_OUT and PITCH_OUT registers provide a tilt angle calculation, based on the accelerometers. The zero reference is the point at which the z-axis faces gravity for a north-east-down (NED) configuration.

$$ROLL_OUT = a \tan\left(\frac{YACCL_OUT}{ZACCL_OUT}\right) = \phi$$
$$PITCH_OUT = a \tan\left(\frac{-XACCL_OUT}{YACCL_OUT \times \sin(\phi) + ZACCL_OUT \times \cos(\phi)}\right)$$

Auxiliary ADC

The AUX_ADC register provides access to the auxiliary ADC input channel. The ADC is a 12-bit successive approximation converter, which has an equivalent input circuit to the one in Figure 14. The maximum input range is +3.3 V. The ESD protection diodes can handle 10 mA without causing irreversible damage. The switch on-resistance (R1) has a typical value of 100 Ω . The sampling capacitor, C2, has a typical value of 16 pF.

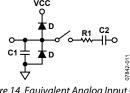


Figure 14. Equivalent Analog Input Circuit (Conversion Phase: Switch Open, Track Phase: Switch Closed)

CALIBRATION

Manual Bias Calibration

The bias offset registers in Table 10 and Table 11 provide a manual adjustment function for the output of each sensor. For example, if GYRO_OFF equals 0x1FF6, the GYRO_OUT offset shifts by -10 LSBs, or -0.125° /sec. The DIN command for the upper byte is DIN = 0x9B1F; for the lower byte, DIN = 0x9AF6.

Table 10. GYRO_OFF

Bits	Description	
[15:13]	Not used.	
[12:0]	Data bits. Twos complement, 0.0125° /sec per LSB. Typical adjustment range = $\pm 50^{\circ}$ /sec.	

Table 11. XACCL_OFF, YACCL_OFF, ZACCL_OFF

Bits	Description
[15:12]	Not used.
[11:0]	Data bits, twos complement 0.6 mg/LSB. Typical adjustment range = ± 1.2 g.

Gyroscope Automatic Bias Null Calibration

Set GLOB_CMD[0] = 1 (DIN = 0xBE01) to execute this function, which measures GYRO_OUT and then loads GYRO_OFF with the opposite value to provide a quick bias calibration. Then, all sensor data resets to zero, and the flash memory updates automatically (50 ms). See Table 12.

Gyroscope Precision Automatic Bias Null Calibration

Set GLOB_CMD[4] = 1 (DIN = 0xBE10) to execute this function, which takes the sensor offline for 30 seconds while it collects a set of GYRO_OUT data and calculates a more accurate bias correction factor. Once calculated, the correction factor loads into GYRO_OFF, all sensor data resets to zero, and the flash memory updates automatically (50 ms). See Table 12.

Restoring Factory Calibration

Set $GLOB_CMD[1] = 1$ (DIN = 0xBE02) to execute this function, which resets each user calibration register (see Table 10 and Table 11) to 0x0000, resets all sensor data to zero, and automatically updates the flash memory (50 ms). See Table 12.

Linear Acceleration Bias Compensation (Gyroscope)

Set $MSC_CTRL[7] = 1$ (DIN = 0xB486) to enable correction for low frequency acceleration influences on gyroscope bias. Note that the DIN sequence also preserves the factory default condition for the data ready function (see Table 17).

OPERATIONAL CONTROL

Global Commands

The GLOB_CMD register provides trigger bits for several useful functions. Setting the assigned bit to 1 starts each operation, which returns to the bit to 0 after completion. For example, set GLOB_CMD[7] = 1 (DIN = 0xBE80) to execute a software reset, which stops the sensor operation and runs the device through its start-up sequence. This includes loading the control registers with their respective flash memory locations prior to producing new data. Reading the GLOB_CMD registers (DIN = 0x3E00) starts the burst mode read sequence.

Table 12. GLOB_CMD

Bits	Description
[15:8]	Not used
[7]	Software reset command
[6:5]	Not used
[4]	Precision autonull command
[3]	Flash update command
[2]	Auxiliary DAC data latch
[1]	Factory calibration restore command
[0]	Autonull command

Internal Sample Rate

The ADIS16300 performs best when the sample rate is set to the factory default setting of 819.2 SPS. For applications that value lower sample rates, the SMPL_PRD register controls the ADIS16300 internal sample (see Table 13), and the following relationship produces the sample rate:

 $t_S = t_B \times N_S + 1$

Bit	Description
[15:8]	Not used
[7]	Time base (t₀) 0 = 0.61035 ms, 1 = 18.921 ms
[6:0]	Increment setting (Ns) Internal sample period = $t_s = t_B \times N_s + 1$

For example, set SMPL_PRD[7:0] = 0x0A (DIN = 0xB60A) for an internal sample period of 6.7 ms (sample rate = 149 SPS). For systems that value lower sample rates, in-system characterization can help determine performance trade-offs.

Power Management

Setting SMPL_PRD \geq 0x0A also sets the sensor in low power mode. In addition to sensor performance, this mode also affects SPI data rates (see Table 2). Two sleep mode options are listed in Table 14. Set SLP_CNT[8] = 1 (DIN = 0xBB01) to start the indefinite sleep mode, which requires \overline{CS} assertion (high to low), reset, or power cycle to wake-up. Set SLP_CNT[7:0] = 0x64 (DIN = 0xBA64) to put the ADIS16300 to sleep for 100 seconds, as an example of the programmable sleep time option.

Table	14.	SLP_	CNT
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Bit	Description
[15:9]	Not used
[8]	Indefinite sleep mode, set to 1
[7:0]	Programmable time bits, 0.5 sec/LSB

Digital Filtering

The signal conditioning circuit of each sensor has a typical analog bandwidth of 350 Hz. A programmable Bartlett window FIR filter provides an opportunity for additional noise reduction on all output data registers. SENS_AVG[2:0] controls the number of taps according to the equation in Table 15. For example, set SENS_AVG[2:0] = 110 (DIN = 0xB806) to establish a 129-tap setting.

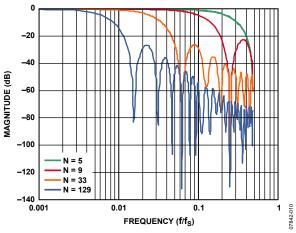


Figure 15. Bartlett Window FIR Frequency Response

Dynamic Range

There are three dynamic range settings for the gyroscope: $\pm 75^{\circ}$ /sec, $\pm 150^{\circ}$ /sec, and $\pm 300^{\circ}$ /sec. The lower dynamic range settings ($\pm 75^{\circ}$ /sec and $\pm 150^{\circ}$ /sec) limit the minimum filter tap sizes to maintain the resolution as the measurement range decreases. The recommended order for programming the SENS_AVG register is upper byte (sensitivity), followed by lower byte (filtering). For example, set SENS_AVG[10:8] = 010 (DIN = 0xB902) for a measurement range to $\pm 150^{\circ}$ /sec.

Table 15. SENS_AVG

Bits	Value	Description	
[15:11]		Not used	
[10:8]		Measurement range (sensitivity) selection	
	100	±300°/sec (default condition)	
	010	$\pm 150^{\circ}$ /sec, filter taps ≥ 4 (Bits[2:0] $\geq 0x02$)	
	001	$\pm 75^{\circ}$ /sec, filter taps ≥ 16 (Bits[2:0] $\geq 0x04$)	
[7:3]		Not used	
[2:0]		Filter tap setting, number of taps	
		$N = 2^{M}+1$ for $M > 0$, $N = 1$ for $M = 0$	

INPUT/OUTPUT FUNCTIONS

General-Purpose I/O

DIO1, DIO2, DIO3, and DIO4 are configurable, general-purpose I/O lines that serve multiple purposes according to the following control register priority: MSC_CTRL, ALM_CTRL, and GPIO_CTRL. For example, set GPIO_CTRL = 0x080C (DIN = 0xB508, then 0xB40C) to set DIO1 and DIO2 as inputs and DIO3 and DIO4 as outputs, with DIO3 set low and DIO4 set high.

Table 16. GPIO_CTRL

Bit	Description
[15:12]	Not used
[11]	General-Purpose I/O Line 4 (DIO4) data level
[10]	General-Purpose I/O Line 3 (DIO3) data level
[9]	General-Purpose I/O Line 2 (DIO2) data level
[8]	General-Purpose I/O Line 1 (DIO1) data level
[7:4]	Not used
[3]	General-Purpose I/O Line 4 (DIO4), direction control
	1 = output, 0 = input
[2]	General-Purpose I/O Line 3 (DIO3), direction control
	1 = output, 0 = input
[1]	General-Purpose I/O Line 2 (DIO2), direction control
	1 = output, 0 = input
[0]	General-Purpose I/O Line 1 (DIO1), direction control
	1 = output, 0 = input

Input Clock Configuration

The input clock configuration function allows for external control over sampling in the ADIS16300. Set GPIO_CTRL[3] = 0 (DIN = 0x0B200) and SMPL_PRD[7:0] = 0x00 (DIN = 0xB600) to enable this function. See Table 2 and Figure 4 for timing information.

Data Ready I/O Indicator

The factory default sets DIO1 as a positive data ready indicator signal. The MSC_CTRL[2:0] register provides configuration options for changing this. For example, set MSC_CTRL[2:0] = 100 (DIN = 0xB404) to change the polarity of the data ready signal for interrupt inputs that require negative logic inputs for activation. The pulse width will be between 100 μ s and 200 μ s over all conditions.

Table 17	. MSC	_CTRI
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Bits	Description	
[15:12]	Not used	
[11]	Memory test (clears on completion)	
	1 = enabled, 0 = disabled	
[10]	Internal self-test enable (clears on completion)	
	1 = enabled, 0 = disabled	
[9]	Manual self-test, negative stimulus	
	1 = enabled, 0 = disabled	
[8]	Manual self-test, positive stimulus	
	1 = enabled, 0 = disabled	
[7]	Linear acceleration bias compensation for gyroscopes	
	1 = enabled, 0 = disabled	
[6]	Linear accelerometer origin alignment	
	1 = enabled, 0 = disabled	
[5:3]	Not used	
[2]	Data ready enable	
	1 = enabled, 0 = disabled	
[1]	Data ready polarity	
	1 = active high, 0 = active low	
[0]	Data ready line select	
	1 = DIO2, 0 = DIO1	

Auxiliary DAC

The 12-bit AUX_DAC line can drive its output to within 5 mV of the ground reference when it is not sinking current. As the output approaches 0 V, the linearity begins to degrade (~100 LSB beginning point). As the sink current increases, the nonlinear range increases. The DAC latch command moves the values of the AUX_DAC register into the DAC input register, enabling both bytes to take effect at the same time.

Table 18. AUX_DAC

Bit	Description
[15:12]	Not used
[11:0]	Data bits, scale factor = 0.8059 mV/code Offset binary format, 0 V = 0 codes

Table 19. Setting AUX_DAC = 1V

DIN	Descripition
0XB0D9	AUX_DAC[7:0] = 0xD9 (217 LSB).
0xB104	AUX_DAC[15:8] = 0x04 (1024 LSB).
0xBE04	$GLOB_CMD[2] = 1.$
	Move values into the DAC input register, resulting in a 1 V output level.

DIAGNOSTICS

Self-Test

Self-test offers the opportunity to verify the mechanical integrity of each MEMS sensor. It applies an electrostatic force to each sensor element, which results in mechanical displacement that simulates a response to actual motion. Table 1 lists the expected response for each sensor, which provides pass/fail criteria. Set MSC_CTRL[10] = 1 (DIN = 0xB504) to run the internal self-test routine, which exercises all inertial sensors, measures each response, makes pass/fail decisions, and reports them to error flags in the DIAG_STAT register. MSC_CTRL[10] resets itself to 0 after completing the routine. MSC_CTRL[9:8] (DIN = 0xB502 or 0xB501) provides manual control over the self-test function. Table 20 gives an example test flow for using this option.

Table 20. Manual Self-Test Example Sequence

DIN	Description
0xB601	SMPL_PRD[7:0] = 0x01, sample rate = 819.2 SPS.
0xB904	SENS_AVG[15:8] = $0x04$, gyro range = $\pm 300^{\circ}$ /sec.
0xB802	SENS_AVG[7:0] = $0x02$, 4-tap averaging filter.
	Delay = 50 ms.
0x0400	Read GYRO_OUT.
0x0600	Read XACCL_OUT.
0x0800	Read YACCL_OUT.
0x0A00	Read ZACCL_OUT.
0xB502	MSC_CTRL[9] = 1, gyroscope negative self-test.
	Delay = 50 ms.
0x0400	Read GYRO_OUT.
	Determine whether the bias in the gyroscope
	output changes according to the expectation set
	in Table 2.
0xB501	MSC_CTRL[9:8] = 01, gyroscope/accelerometer positive self-test.
	Delay = 50 ms.
0x0400	Read GYRO_OUT.
0x0600	Read XACCL_OUT.
0x0800	Read YACCL_OUT.
0x0A00	Read ZACCL_OUT
	Determine whether the bias in the gyroscope and
	accelerometers changed according to the expect-
	ation set in Table 2.
0xB500	$MSC_CTRL[15:8] = 0x00.$

Zero motion provides results that are more reliable. The settings in Table 20 are flexible and provide opportunity for optimization around speed and noise influence. For example, lowering the filtering taps enables lower delay times but increases the opportunity for noise influence.

Memory Test

Setting MSC_CTRL[11] = 1 (DIN = 0xB508) does a check-sum verification of the flash memory locations. The pass/fail criteria load into the DIAG_STAT[6] register.

Status

The error flags provide indicator functions for common system level issues. All of the flags clear (set to 0) after each DIAG_STAT register read cycle. If an error condition remains, the error flag returns to 1 during the next sample cycle. DIAG_STAT[1:0] does not require a read of this register to return to zero. If the power supply voltage goes back into range, these two flags clear automatically.

Table 21. DIAG_	STAT Bit Descriptions

Table 21. DIAG_STAT bit Descriptions		
Bit	Description	
[15]	Z-axis accelerometer self-test failure	
	1 = error condition, 0 = normal operation	
[14]	Y-axis accelerometer self-test failure	
	1 = error condition, 0 = normal operation	
[13]	X-axis accelerometer self-test failure	
	1 = error condition, 0 = normal operation	
[12:11]	Not used	
[10]	Gyroscope self-test failure	
	1 = error condition, 0 = normal operation	
[9]	Alarm 2 status	
	1 = active, 0 = inactive	
[8]	Alarm 1 status	
	1 = active, 0 = inactive	
[7]	Not used	
[6]	Flash test, check-sum flag	
	1 = failure, 0 = normal operation	
[5]	Self-test diagnostic error flag	
	1 = error condition, 0 = normal operation	
[4]	Sensor overrange	
	1 = error condition, 0 = normal operation	
[3]	SPI communications failure	
	1 = error condition, 0 = normal operation	
[2]	Flash update failed	
	1 = error condition, $0 = $ normal operation	
[1]	Power supply above 5.25 V	
[0]	1 = power supply ≥ 5.25 V, 0 = power supply ≤ 5.25 V	
[0]	Power supply below 4.75 V 1 = power supply \leq 4.75 V, 0 = power supply \geq 4.75 V	
	$1 - power suppry \ge 4.75$ v, $0 - power suppry \ge 4.75$ v	

Alarm Registers

The alarm function provides monitoring for two independent conditions. The ALM_CTRL register provides control inputs for data source, data filtering (prior to comparison), static comparison, dynamic rate-of-change comparison, and output indicator configurations. The ALM_MAGx registers establish the trigger threshold and polarity configurations.

Table 25 gives an example of how to configure a static alarm.

The ALM_SMPLx registers provide the numbers of samples to use in the dynamic rate-of-change configuration. The period equals the number in the ALM_SMPLx register, multiplied by the sample period time, established by the SMPL_PRD register. See Table 26 for an example of how to configure the sensor for this type of function.

Table 22. ALM_MAG1, ALM_MAG2

Bit	Description
[15]	Comparison polarity
	1 = greater than, $0 =$ less than
[14]	Not used
[13:0]	Data bits that match the format of the trigger source selection

Table 23. ALM_SMPL1, ALM_SMPL2

Bit	Description
[15:8]	Not used
[7:0]	Data bits: number of samples (both $0x00$ and $0x01 = 1$)

Table 24. ALM_CTRL Bit Designations

Bits	Value	Description
[15:12]		Alarm 2 source selection
	0000	Disable
	0001	Power supply output
	0010	Gyroscope output
	0011	Not used
	0100	Not used
	0101	X-axis accelerometer output
	0110	Y-axis accelerometer output
	0111	Z-axis accelerometer output
	1000	Gyroscopes temperature output
	1001	X-axis inclinometer output
	1010	Y- axis inclinometer output
	1011	Auxiliary ADC input
[11:8]		Alarm 1 source selection (same as Alarm 2)
[7]		Rate of change (ROC) enable for Alarm 2 1 = rate of change, 0 = static level
[6]		Rate of change (ROC) enable for Alarm 1 1 = rate of change, 0 = static level
[5]		Not used
[4]		Comparison data filter setting ¹ 1 = filtered data, 0 = unfiltered data
[3]		Not used
[2]		Alarm output enable 1 = enabled, 0 = disabled
[1]		Alarm output polarity 1 = active high, 0 = active low
[0]		Alarm output line select 1 = DIO2, 0 = DIO1

Table 25. Alarm Configuration Example 1

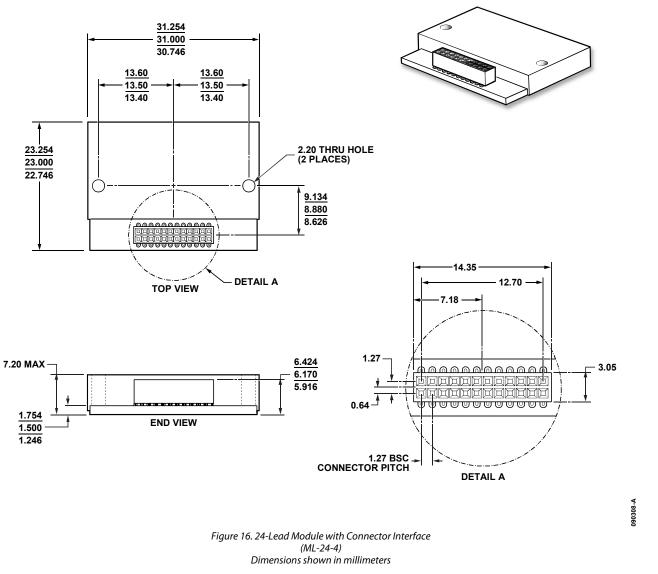
DIN	Description
0xAF55	ALM_CTRL = 0x5517.
0xAE17	Alarm 1 input = XACCL_OUT.
	Alarm 2 input = XACCL_OUT.
	Static level comparison, filtered data.
	DIO2 output indicator, positive polarity.
0xA783	ALM_MAG1 = 0x8341.
0xA641	Alarm 1 is true if XACCL_OUT > 0.5 g.
0xA93C	ALM_MAG2= 0x3CBF.
0xA8BF	Alarm 2 is true if XACCL_OUT < $-0.5 g$.

Table 26. Alarm Configuration Example 2

DIN	Description
0xAF76	ALM_CTRL = 0x7687.
0xAE87	Alarm 1 input = ZACCL_OUT.
	Alarm 2 input = YACCL_OUT.
	Rate of change comparison, unfiltered data.
	DIO2 output indicator, positive polarity.
0xB601	$SMPL_PRD = 0x0001.$
	Sample rate = 819.2 SPS.
0xAB08	$ALM_SMPL1 = 0x0008.$
	Alarm 1 rate of change period = 9.77 ms.
0xAC50	ALM_SMPL2= 0x0050.
	Alarm 2 rate of change period = 97.7 ms.
0xA783	ALM_MAG1 = 0x8341.
0xA641	Alarm 1 is true if XACCL_OUT > 0.5 g.
0xA93C	ALM_MAG2= 0x3CBE.
0xA8BE	Alarm 2 is true if XACCL_OUT < $-0.5 g$.

¹ Incline outputs always use filtered data in this comparison.

OUTLINE DIMENSIONS



ORDERING GUIDE

Model	Temperature Range	Package Description	Package Option
ADIS16300AMLZ ¹	–40°C to +85°C	24-Lead Module with Connector Interface	ML-24-4

 1 Z = RoHS Compliant Part.

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